

# Fabrication, Mechanical Characterization, and Selection of Hybrid Composites by TOPSIS

Raffi Mohammed, B Ramgopal Reddy, K Sridhar, Aluri Manoj

**Abstract:** Fiber reinforced polymer composites are utilized in an assortment of use in light of their numerous favorable circumstances, for example, moderately ease of creation, simple to manufacture and better quality analyze than flawless polymer resins. Fortification in a polymer is neither manufactured nor normal. A manufactured fiber, for example, carbon, glass and so forth has high explicit quality however their fields of use are constrained because of the greater expense of generation. As of late, there is an expanded enthusiasm for regular filler based composites because of their numerous points of interest. In this association, an examination has been done to improve the usage of Bagasse cinder /coal fly fiery debris/coal powder as filler material alongside glass fiber as fortification for the readiness of epoxy based mixture composites. The target of the present investigation is to examine the mechanical properties of composite reinforced with epoxy glass based on hybrid epoxy. The impact of filler material and fiber impact stacking on mechanical properties like elasticity, ILSS, tensile modulus, flexural quality, hardness and impact strength of composites is examined. Multi-Criteria basic leadership access named TOPSIS is likewise used to choose the good option from a lot of choices dependent on various characteristics (mechanical attributes).

**Keywords:** TOPSIS, ILSS, Fiber Reinforced polymers, Flexural Strength, Tensile Strength.

## I. INTRODUCTION

New composites development (or) even improving the existing ones has become a very challenging task for the research scholars in the domain of material science. On par with conventional materials, Polymer Matrix Composites (PMC) are very convenient in a way to displace traditional metallic materials as they are easy for production, process, and reduction in cost, etc. From all the thermosetting polymers, the epoxy resins are favorable for applications of high performance as they are having good thermal and mechanical inheritance and corrosion and chemical resistance is also high but on curing it has low shrink and under different conditions it can be processed [1]. One of the approachable way to establish a present day classification of materials of

polymer structural is the modification of polymer matrix. Glass and carbon fiber reinforcement enhance the polymer residences [2]. Monolithic substances fail to end result in the astonishing combo of mechanical and physical attributes [3, 4]. Void content, orientation and the distribution of fiber, trait and fiber type will be identifying elements for the mechanical houses of the composites with fiber. On the different hand interphase mechanism of the load transfer and interfacial bonds aspects additionally play an vital role [5]. Significant discount in the fee of processing and the mechanical properties advancement of the composites which are based on epoxy with micro fillers addition and these were acting as extra reinforcing components. The used filler material size, weight percentage, and kind are important elements for composites attributes [6,7]. The mechanical attributes assessment and mechanical characterization of hybrid composites based totally on epoxy which is reinforced with E-glass with identical fillers weight percent like coal fly ash/coal powder/ Bagasse ash are no longer studied yet. In this paper hybrid composites based on epoxy are filled with E-glass reinforced particulates are organized to find out about their mechanical properties with the filler weight expansion of 5wt% and 10wt%. In case of including modifiers to epoxy, some enhancement in mechanical attributes can be observed.

TOPSIS is a extraordinary precept method in a way to apprehend effects from defined collected picks structured on synchronous minimization of separation from a factor which is perfect and separation increase from point of nadir. Many kinds of lookup had related TOPSIS numerous applications. Determination of material for bike chain fabric in Indian state of affairs utilising by means of drawing near through MADDM has been examined by way of Singh et al. [8] and reasoned that both the strategies TOPSIS and MADDM are consumer amicable for positioning the limitations. Multi-criteria basic leadership and vulnerability examination for materials viewed to determine the ecologically cognizant structure by Huang et al. [9]. TOPSIS approach well-known shows a sensible execution for getting answer has been reported, and the technique of entropy introduces architect's both chief's on price choice either ecological effect and weight uncertainties have been shown effectively. Choice of perfect refinement situations to accomplish most intense pliable attributes of Al-15%Mg<sub>2</sub>Si composite which is structured on the approach of TOPSIS and noticed that the TOPSIS approach is been seen as a lifelike methodology for taking care of cloth choice trouble every time the precise execution value determinations are accessible has been reported by means of Khorshid et al.[10]. TOPSIS method and evaluation of records envelopment has been used with the aid of Ghaseminejad et al.

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[11] for settling adaptable narrows shape format and observed that this method is beneficial for making, introductory design, producing opening diagram options and assessing them. TOPSIS-AHP-technique which is combines primarily based on the methodology for non-customary machining forms dedication is examined through Chakladar and Chakraborty [12] and furthermore accommodates the shape alongside with TOPSISAHP-strategy advancement hooked up on a master framework which can robotize the procedure of primary leadership along with the help of a graphical UI as nicely as visible guides. Multi-criteria simple management method (ANP TOPSIS) to determine vendors in Iran's automobile industry has been regarded by Shahroudi and Rouydel [13]. Lin et al. [14] meditated on the procedure of client-directed object configuration with the aid of using TOPSIS and AHP methodologies and conclusions demonstrates the recommended methodology suits to assist architects to intentionally assume about vital diagram facts and accurately determine the key structure ambitions and perfect utilized choices. Multi-criteria basic management (MCDM) way to deal with consider the telephone phone preferences in regard to the clients' inclinations request by means of utilizing TOPSIS method has examined once more by using Isiklar and Buyukozkan [15].

## II. MATERIALS AND APPROACHES

### 2.1 Materials needed

Coal fly ash, Bagasse ash, and coal powder particles are utilized as modifiers. Bagasse ash and Bagasse fiber are gathered from the KCP sugars and Industries Corporation Limited, Vuyyuru. Coal fly ash and coal powder are gathered from the NTTPS, Vijayawada. Gathered Bagasse ash, coal powder and fly ash are restored in the oven at a temperature of 105 °C dampness evacuations, afterward sieved to normal size of 70-80µm. E-Glass fiber for fortification, hardener (HY-951) and the epoxy resin Araldite (LY-556) is provided by Kotson building company, Guntur.

### 2.2 Composite fabrication without filler material

Fiber heaps are sliced from the glass fiber material to the required size. Fiber piles of suitable quantities were taken: 8 for every composite. At that point the fibers were gauged; likewise, the hardeners and resin were gauged. Hardener and Epoxy were blended in a bowl by utilizing stirrer. In order to avoid air pockets care has been taken. If the air bubbles were caught in the matrix then it may result in the material failure. Consequent creation process includes keeping a discharging film on the shape surface at first. Then a covering of polymer was attached on sheets and fiber ply of one type was put and finished suitable rolling and then the resin was connected again, fiber ply of a different sort was rolled by keeping next to it. Utilizing tube-shaped mellow steel pole rolling was finished. Until eight exchanging fibers have been laid this system was rehashed. A polymer covering is done on the top point of the last ply which serves to assure a decent completion of the surface. At last light rolling was completed by keeping a discharging sheet on the main and then a weight of 20 kgf was connected with composite . It was left ideal for

72 hrs for permitting adequate time to restore and consequent solidifying.

### 2.3 Fabrication of hybrid composites (with fillers)

Hand lay-up technique has been utilized to manufacture composite of E-Glass which is reinforced with epoxy primarily based particulate filled. Particulate fillers used are Bagasse/ coal fly ash/ coal powder ash with 10wt%, 5wt% of creation. The composition and designation of hardener, Fillers, Glass fiber and Epoxy are fixed and appeared in Table 3.1. Epoxy is dried at 1050C in an oven for 2 hours in advance of blending with fillers. At that point epoxy is covered with fillers and blended using a wood stirrer or round glass for thirty minutes in advance of the blending of hardener in addition hardener is blended, a comparable methodology to manufacture composite except any filler and subsequent to the introduction of fillers composites. To spread the resin constantly mild metal roller was proceeded onward each layer of glass fiber and additionally to expel air captured interior the composite. This approach was once rehashed up to eight layers of the glass fiber and after that composites are restored at environmental temperature for 72hours.

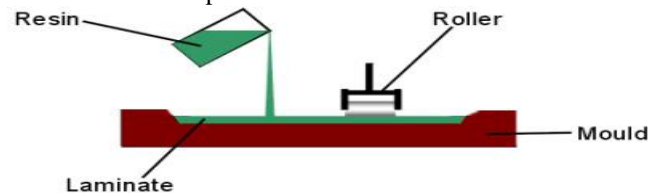


Fig 2.1 Hand layup process

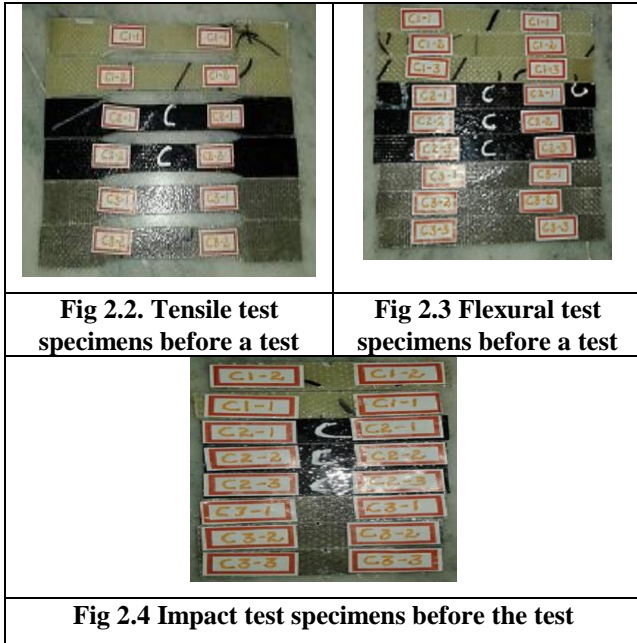
**Table 2.1. Composite constituents & weight percentages**

Designation	Composition
C7	10 wt % Bagasse Ash +50 wt % Glass Fiber + 40 wt % Epoxy Resin
C6	5 wt % Bagasse Ash + 50 wt % Glass Fiber + 45 wt % Epoxy Resin
C5	10 wt % Coal Fly Ash +50 wt % Glass Fiber + 40 wt % Epoxy Resin
C4	5 wt % Coal Fly Ash +50 wt % Glass Fiber + 45 wt % Epoxy Resin
C3	10 wt % Coal Powder +50 wt % Glass Fiber + 40 wt % Epoxy Resin
C2	5 wt % Coal Powder +50 wt % Glass Fiber + 45 wt % Epoxy Resin
C1	50 wt % Glass Fiber + 50 wt % Epoxy Resin

### 2.4 Preparation of the specimen

E-Glass reinforced particulate filled epoxy based half and half composites which were fabricated from molds were taken out. According to ASTM guidelines, they were sliced to the ideal measurements samples from the chunks of composite for the portrayal of mechanical (for example sway tests, flexural, Malleable test) by utilizing hack saw and different apparatuses available in the building workshop different examples of sizes and shapes appear as follows.





**Fig 2.2. Tensile test specimens before a test**

**Fig 2.3 Flexural test specimens before a test**

**Fig 2.4 Impact test specimens before the test**

**2.5 Details of the Material Test**

**2.5.1 Tensile modulus and tensile strength**

Material tensile quality is the most extreme measure of the tensile pressure which takes before failure. For tensile test normally utilized example is pooch bone sort. The load is applied both the finishes of the sample during the test, where the applied load is uni-axial. As per ASTM D638 (TENSILE) the sample dimensions are decided. Run the mill focal points when testing a material incorporate Ultimate Tensile Strength (UTS) or pinnacle pressure; Offset Yield Strength (OYS) just beyond the beginning of lasting misshapening where a point is represented; and Rupture (R) in turn crack point where example gets isolated into pieces. Using Universal testing machine (UTM) the tensile test is performed Instron 1195(40-ton capacity) and conclusions are broke down to figure tensile quality of samples of a composite. Quality of tensile is determined by separating heap at the break by first least zone of a cross-section. In Mega Pascal's (MPa) Outcome is communicated.

**2.5.2 Interlaminar shear strength and Flexural strength**

Flexural strength is characterized as a capacity of the materials to oppose disfigurement beneath the burden. On the samples of composite short beam shear (SBS), tests are performed. It is a three-point twist test, where for the most part advances failure by between laminar shear. By using UTM (limit 60T) according to ASTM D790 (FLEXURAL) this test is directed. In the figure, the stacking course of action has appeared. Flexural quality is communicated as modulus of rupture (MR) in psi(MPa). Flexural MR is around 10 to 20 percent of compressive quality relying upon the volume, size, and sort, of course, total utilized. Anyway, the good relationship for explicit materials is acquired by research facility tests for and blend structure and given materials.

$$\text{Flexural Strength} = 3PL^2/bd^2$$

The ILSS equation is  $ILSS = 3P/4bd$   
Here, the maximum load connected is p, the sample width is b and the sample thickness is d. To ascertain the flexural quality too a similar estimation of p is utilized. A range of 52mm is utilized for acquiring both Flexural strength and ILSS.

**2.5.3 Impact strength**

The ability of the material to cope with an abruptly connected load; is communicated regarding vitality is the impact strength. Charpy impact strength test or Izod impact test regularly estimated. On samples of the composite low speed instrumented impact tests were completed. Tests are performed according to ASTM D 256 utilizing an impact analyzer (Fig 3.9). The indent impact strength of the material is found using Izod/Charpy impact test machine by breaking the V-scored example using a pendulum hammer, estimating the spent vitality, and relating that to the cross segment of the composite sample. For ASTM D 256 is 64mm x 12.7mm x 3.2 mm and the profundity under the indent is 10.2 mm is the standard sample.

**2.6 Technique for Order Preference by Similarity to Ideal Solution Ranking:**

To quantify the closeness to the perfect arrangement, Technique for Order Preference by Similarity to Ideal Solution (TOPSIS) is executed. An essential idea for this strategy is that, from positive they picked option ought to have the most limited separation for perfect arrangement and from negative the most distant separation for the perfect arrangement. The positive perfect arrangement is the composition of good execution esteems illustrated (in decision matrix) by several options for every trait. The composite of the most exceedingly terrible presentation esteems is a negative-perfect arrangement. The means required for ascertaining the TOPSIS esteems are as per the following [29]:

**Stage 1**

This progression includes the advancement of a matrix group. The matrix row is assigned for one option and every section to one quality. This matrix is named as a decision matrix (D). The decision matrix can be communicated as :

$$\begin{matrix} A1 \\ A2 \\ A3 \\ Ai \\ A5 \end{matrix} \begin{bmatrix} X11 & X12 & \dots & X1j & X1n \\ X21 & X22 & \dots & X2j & X2n \\ \dots & \dots & \dots & \dots & \dots \\ Xi1 & Xi2 & \dots & Xij & Xin \\ Xm1 & Xm2 & \dots & Xmj & Xmn \end{bmatrix}$$

**Stage 2**

The standardized decision matrix is calculated. The standardized esteem  $r_{ij}$  is determined as pursues:

$$r_{ij} = x_{ij} \sqrt{\sum_{i=1}^m x_{ij}^2} \quad i=1, 2, \dots, m \text{ and } j = 1, 2, \dots, n.$$

**Stage 3**

A weighted standardized decision matrix is calculated. The weighted standardized esteem is determined as pursues:

$$V_{ij} = r_{ij} \times W_j ; i=1, 2,.. m \text{ and } j = 1, 2, \dots, n.$$

# Fabrication, Mechanical Characterization, and Selection of Hybrid Composites by TOPSIS

$W_j$  weight of the  $j^{th}$  criterion or attribute

$$\text{and } \sum_{j=1}^n W_j = 1.$$

## Stage 4

Find the ideal ( $A^*$ ) and negative ideal ( $A^-$ ) solutions.

$$A^* = \{(\max_i v_{ij} | j \in C_b), (\min_i v_{ij} | j \in C_c)\} = \{v_j^* | j=1,2,\dots,m\}$$

$$A^- = \{(\min_i v_{ij} | j \in C_b), (\max_i v_{ij} | j \in C_c)\} = \{v_j^- | j=1,2,\dots,m\}$$

## Stage 5

By using m-dimensional Euclidean distance detachment measures are calculated. The detachment measures of every option from the negative ideal solution and positive ideal solution separately, are as per the following:

$$S_i^- = \sqrt{\sum_{j=1}^m (v_{ij} - v_j^-)^2}, j = 1,2,\dots,m$$

$$S_i^* = \sqrt{\sum_{j=1}^m (v_{ij} - v_j^*)^2}, j = 1,2,\dots,m$$

## Stage 6

Figure the ideal solution's relative closeness. The overall closeness option concerning is characterized as pursues:

$$RC_i^* = \frac{S_i^-}{S_i^* + S_i^-}, i = 1,2,m$$

## Stage 7

Rank the inclination order.

## III. RESULTS AND DISCUSSION

### 3.1. Characterization of the Composite

Composites mechanical properties and the portrayal of composites uncovers such incorporation of any filler particulate has a solid impact on the mechanical properties of composites and physical properties of composites. Changed

estimations of composite properties under this examination are displayed then thought about against composite of unfilled glass epoxy in Table 3.1.

Composite Designation	Tensile Strength (MPa)	ILSS (GPa)	Flexural Strength (MPa)	Hardness (Hb)	Impact Strength J/m <sup>2</sup>	Tensile Modulus (GPa)
C7	159.91	7.84	189.58	85.66	4	5.688
C6	175.71	6.542	214.43	89	3.6	9.562
C5	175.133	24.247	693.07	56	4	7.823
C4	234.47	9.696	525.42	54	1.34	5.850
C3	181.786	25.188	750.54	61	5.6	5.665
C2	249	9.013	488.4	58	1.85	5.225
C1	252.18	12.30155	666.58	47	2.2	6.292

### 3.2 Tensile Modulus & Tensile Strength of Composites

The test results for rigid qualities and Tensile Modulus appear in Table 3.1. C1 composite (50 wt % Glass Fiber + 50 wt % Epoxy Resin) Exhibited the most extreme tensile strength of 252.18MPa and C6 composite (5 wt % Bagasse Ash +45 wt %+50 wt % Glass Fiber+ Epoxy Resin) exhibited a greatest tensile modulus of 9.562GPa The decrease in Tensile Modulus & Tensile strength of composites is a result of

- Pores presence between filler particles and the matrix at the interface. The interfacing attachment with particulate filled composites. might be too feeble to even think about transferring the ductile stress.
- In the matrix base stress concentration results because of the unpredictable formed particulates.

**3.3. ILSS Properties and Flexural strength**

The results of the test to flexural qualities appear in Table 3.1. It is observed that C3 Composite (10 wt % Coal

1. The inconsistency of the epoxy matrix and the particulates, prompting less interfacial holding.
2. The fewer estimations flexural quality may likewise be credited from one fiber to one fiber connection, scattering and voids issues

Anyway, it additionally relies upon different criteria such as size, shape, type and stacking on filler for the material.

**3.4 Composites Hardness**

From Table 3.1 it is observed at C6-5 wt % Bagasse Ash +50 wt % Glass Fiber+45 wt % Epoxy Resin showed hardness of 89 which is greatest and showed diminished hardness for residual composites. Increased hardness was found by adding filler. This one is on the grounds that amid the compressive stacking in hardness test the support stage (for example glass fiber and filler) and matrix stage are squeezed closely firm so that the interface can exchange weight all the more successfully which shows improvement in hardness

**3.5. Composite Impact Strength**

Various composites impact energy estimations recorded amid the Table 3.1 consists of impact tests. It demonstrates that the protection from impact stacking composites of glass epoxy advances with particulate filler expansion in

Powder+50 wt % Glass Fiber+ 40 wt % Epoxy Resin) Flexural and ILSS maximum as appeared in the Graph.

It can have two explanations behind the Flexural quality properties decrease of the unfilled composites contrasted Composite C3. The quality of impact of C3 composite (10 wt % Coal Powder +50 wt % Glass Fiber+ 40 wt % Epoxy Resin) is 5.6J/m2 and diminished for other unfilled and filled composites. Impact quality reduction is because of lessening in the energy engrossing limit with the expansion of filler, the decline in the energy retaining limit in the composite is because of the reason that the portability of polymer chain is obliged using filler content, it decreases the capacity to distort uninhibitedly and material is made less bendable.

**3.6. Composites ranking using TOPSIS Method**

Every composite material looks at dependent on TOPSIS strategy and done with the ranking. The weight normalization matrix, normalization matrix, decision matrix, perfect positive and perfect negative arrangement, partition measure; relative closeness esteem and positioning are organized in Tables beneath. At last, the positioning of various composite dependent on their attributes is being appeared .

**STAGE-1:** This progression includes the advancement of a matrix group. The matrix row is assigned for one option and every section to one quality. This matrix is named as a decision matrix

Table 3.2: Decision Matrix (D) of Composites						
Composite Designation	DECISION MATRIX(D)					
	F.S	T.M	H	IS	ILSS	T.S
C7	189.58	5.688	85.66	4	7.84	159.91
C6	214.43	9.562	89	3.6	6.542	175.71
C5	693.07	7.823	56	4	24.247	175.133
C4	525.42	5.85	54	1.34	9.696	234.47
C3	750.54	5.665	61	5.6	25.188	181.786
C2	488.4	5.225	58	1.85	9.013	249
C1	666.58	6.292	47	2.2	12.30155	252.18

**STEP-2:** The standardized decision matrix is calculated. The standardized esteem  $r_{ij}$  is determined as pursues:

Table 3.3: Normalized Matrix						
Composite Designation	NORMALIZED MATRIX					
	F.S	T.M	H	IS	ILSS	T.S
C7	0.131307903	0.31887239	0.48964854	0.430386032	0.192821129	0.291597459
C6	0.14851964	0.5360509	0.50874060	0.38734742	0.160897427	0.32040891

## Fabrication, Mechanical Characterization, and Selection of Hybrid Composites by TOPSIS

<b>C5</b>	0.48003781	0.43856165	0.32010644	0.43038603	0.59634361	0.31935674
<b>C4</b>	0.36391918	0.32795419	0.30867407	0.14417932	0.23846858	0.42755835
<b>C3</b>	0.51984298	0.31758299	0.34868737	0.60254044	0.61948706	0.33148856
<b>C2</b>	0.33827819	0.29291635	0.33153882	0.19905354	0.22167051	0.45405395
<b>C1</b>	0.46169016	0.35273296	0.26866076	0.23671231	0.30255086	0.45985271

**STAGE-3:** Weighted standardized decision matrix is calculated. The weighted standardized esteem is determined as pursues:

<b>Table 3.4: Weight Normalized Matrix (W)</b>						
<b>Composite Designation</b>	<b>WEIGHT NORMALIZED MATRIX</b>					
	<b>F.S</b>	<b>T.M</b>	<b>H</b>	<b>IS</b>	<b>ILSS</b>	<b>T.S</b>
<b>C7</b>	0.021884651	0.05314539	0.08160809	0.071731005	0.032136855	0.04859957
<b>C6</b>	0.024753274	0.08934182	0.0847901	0.064557905	0.026816238	0.05340148
<b>C5</b>	0.080006302	0.07309361	0.053351074	0.071731005	0.099390602	0.05322612
<b>C4</b>	0.060653197	0.05465903	0.051445679	0.024029887	0.039744763	0.07125972
<b>C3</b>	0.086640498	0.0529305	0.058114563	0.100423407	0.103247844	0.05524809
<b>C2</b>	0.056379699	0.04881939	0.05525647	0.03317559	0.036945086	0.07567565
<b>C1</b>	0.076948361	0.05878882	0.044776795	0.039452053	0.050425144	0.07664211

**STEP-4:** Find the ideal ( $A^*$ ) and negative ideal ( $A^-$ ) solutions.

<b>Table 3.5: Best &amp; Worst Solutions</b>						
<b>Ideal Solution</b>	<b>BEST &amp; WORST SOLUTIONS</b>					
	<b>F.S</b>	<b>T.M</b>	<b>H</b>	<b>IS</b>	<b>ILSS</b>	<b>T.S</b>
<b>Negative Ideal Solution(<math>A^-</math>)</b>	0.02188	0.04882	0.04478	0.02403	0.02682	0.0486
<b>Positive Ideal Solution(<math>A^*</math>)</b>	0.08664	0.08934	0.08479	0.10042	0.10325	0.07664

**STEP-5:** By using m-dimensional Euclidean distance detachment measures are calculated. The detachment measures of every option from the negative ideal solution and positive ideal solution separately, are as per the following:

<b>Table 3.6: Attributes Separation Measures</b>		
<b>Composite Designation</b>	<b>Attributes Separation Measures</b>	
	<b>S<sup>-</sup></b>	<b>S<sup>*</sup></b>
<b>C7</b>	0.060652058	0.046766812
<b>C6</b>	0.07011881	0.030354291
<b>C5</b>	0.107724731	0.081103152
<b>C4</b>	0.047564683	0.063412028
<b>C3</b>	0.126925152	0.055408491
<b>C2</b>	0.047108769	0.060787206
<b>C1</b>	0.068653513	0.060121454

**STEP6&7:** Figure the ideal solution's relative closeness. The alternative relative closeness  $A_i$  in regard with  $A^*$  is described as follows and Rank the preference order.

Table 3.6: Relative Closeness(C1*) & Composite Ranking®		
Composite Designation	Relative closeness & Composite Ranking	
	C1*	R
C7	0.435368682	4
C6	0.30211361	7
C5	0.42950835	5
C4	0.571399417	1
C3	0.303885176	6
C2	0.563387151	2
C1	0.466872215	3

#### IV. CONCLUSION

##### 4.1 Conclusion of the Project

The trial examination on fiber impact stacking and content of filler on fiber mechanical conduct fortified composites of epoxy were directed. Attributes, for example, Impact energy, quality of flexural, the quality of tensile, were assessed. Trials push us to the accompanying ends acquired from through this examination:

1. The effective creations of another class of hybrid composites of epoxy fortified are loaded up with different fillers with glass fiber and Bagasse ash /coal fly ash /coal powder has been finished.

2. Filler material impact on mechanical properties has been examined after completion of Mechanical characterization.

3. Composite selection has been finished by TOPSIS dependent on mechanical attributes.

Conceivable utilization of the composites, for example, channels conveying coal dust, modern fans, helicopter fan sharp edges, desert structures, minimal effort lodging and so forth is suggested. In any case, this examination can be additionally stretched out in eventual to new composite sorts utilizing other fillers/inorganic materials and the subsequent test discoveries can be comparatively broke down.

##### 4.2 Future scope

There is a ample extension for upcoming researchers to investigate this research zone. This work can be also be stretched out to ponder different parts of those composites like utilization of different filler potential for assessment of their mechanical and advancement of hybrid composites and disintegration conduct and the subsequent test discoveries can be comparatively broke down.

#### REFERENCES

1. Raffi Mohammed etc., "Effect of Epoxy modifiers (Bagasse fiber / Bagasse ash / Coal powder /Coal Fly ash) on mechanical properties of Epoxy / Glass fiber hybrid composites," *International Journal of Applied Engineering Research.*, ISSN 0973-4562; 2015: Vol. 10 No.24, pp- 45625-45630
2. A. Yasmin and I. M. Daniel, "Mechanical and Thermal Properties of GraphitePlatelet/EpoxyComposites," *Polymer.*, Vol.45,No.24,2004, pp.8211-8219.
3. B. Kotiveerachari, A. Chennakesava Reddy, "Interfacial effect on the fracture mechanism in GFRP composites," *CEMILAC Conference, Ministry of Defence, India.*, 20-21st August 1999, pp. 85-87.
4. A. Chennakesava Reddy, M. Vidya Sagar, "Two-dimensional theoretical modelling of anisotropic wear in carbon/epoxy FRP

5. Cs. Varga, N. Miskolczi, L. Bartha and G. Lipoczi., "Improving the Mechanical Properties of Glass-Fibre-Rein- forced Polyester Composites by Modification of Fibre Surface," *Materials and Design.*Vol. 31,2010, pp. 185- 193.
6. R. Sateesh Raja, K. Manisekar, and V. Manikandan., "Effect of fly ash filler size on mechanical properties of polymer matrix composites," *International Journal of Mining, Metallurgy and Mechanical Engineering.*(1-1), 2013,pp.34-37.
7. K. Devendra, and T. Rangaswamy., "Strength characterization of E-glass fiber reinforced epoxy composites with filler materials," *Journal of Minerals, Material Characterization and Engineering.*(1), 2013pp.353-357.
8. Singh, H., & Kumar, R., "Selection of Material for Bicycle Chain in Indian Scenario using MADM Approach", *Proceedings of the World Congress on Engineering*, Vol. 3,2012.
9. Huang, H., Zhang, L., Liu, Z., & Sutherland, J. W. "Multi-criteria decision making and uncertainty analysis for materials selection in environmentally conscious design," *The International Journal of Advanced Manufacturing Technology*;2011: 52(5-8), pp.421-432.
10. Khorshidi R., Hassani A., Honarbakhsh R.A. and Emamy M., "Selection of an optimal refinement condition to achieve maximum tensile properties of Al-15% Al- 15%Mg2Si composite based on TOPSIS method," *Materials & Design*, 42, 2013, pp. 442-450.
11. Ghaseminejad A., Navidi H., & Bashiri M., "Using Data Envelopment Analysis and TOPSIS method for solving flexible bay structure layout," *International Journal of Management Science and Engineering Management*,1(6), 2011, pp.49-57.
12. Chakladar N. D. & Chakraborty S., "A combined TOPSIS-AHP-method-based approach for non-traditional machining processes selection," *Proceedings of the Institution of Mechanical Engineers, Part B: Journal of Engineering Manufacture*,222(12), 2008,pp.1613-1623.
13. Shahroudi K., & Rouydel H., "Using a multi-criteria decision making approach (ANP-TOPSIS) to evaluate suppliers in Iran's auto industry," *International Journal of Applied*, 2(2), 2012, pp. 37-48.
14. Lin M. C., Wang C. C., Chen M. S., & Chang C. A., "Using AHP and TOPSIS approaches in customer-driven product design process," *Computers in Industry*,59(1), 2008, pp.17-31.
15. Isiklar G., & Buyukozkan G. "Using a multi-criteria decision making approach to evaluate mobile phone alternatives," *Computer Standards & Interfaces*,29(2), 2007, pp.265-274.

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## Fabrication, Mechanical Characterization, and Selection of Hybrid Composites by TOPSIS



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